

**GUJARAT TECHNOLOGICAL UNIVERSITY**  
**BE - SEMESTER- VIII • EXAMINATION – SUMMER 2015**

**Subject Code: 180105****Date: 05/05/2015****Subject Name: High Speed Aerodynamics and Experimental Techniques****Time: 10.30 AM to 01:00 PM****Total Marks: 70****Instructions:**

1. Attempt all questions.
2. Make suitable assumptions wherever necessary.
3. Figures to the right indicate full marks.

- Q.1** (a) Explain with neat sketch physical effects characteristics of hypersonic flow. **07**  
 (b) Derive hypersonic shock relations in terms of the hypersonic similarity parameters for pressure ratio term. **07**
- Q.2** (a) Apply centrifugal force correction to Newtonian flow theory. **07**  
 (b) Derive Newtonian sine squared law for pressure coefficient. **07**
- OR**
- (b) What are the design considerations for supersonic aircraft? **07**
- Q.3** (a) Derive modified Newtonian equation for hypersonic flow. **07**  
 (b) What is wind tunnel? Explain construction of subsonic open type wind tunnel with neat sketch. **07**
- OR**
- Q.3** (a) What is Newtonian theory? Derive  $\beta = \theta$  for hypersonic flow. **07**  
 (b) Explain tangent cone method with neat sketch. **07**
- Q.4** (a) Draw and explain shock wave & Mach wave patterns for Supersonic and Hypersonic flow for Airfoil. **07**  
 (b) Derive the equation Rankine Hugoniot for oblique shock wave. **07**
- OR**
- Q.4** (a) Write a short note on aerodynamic heating for hypersonic flow. **07**  
 (b) Explain solid blockage and wave blockage for wind tunnel. **07**
- Q.5** (a) Explain with neat sketch swept wing. **07**  
 (b) Define Supersonic and Hypersonic flow. Draw and explain entropy layer for airfoil in hypersonic flow. **07**
- OR**
- Q.5** (a) Write a short note on Delta wing with neat sketch. **07**  
 (b) Derive  $L/D = \cot \alpha$  for Flat plate. **07**

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